

The image features a detailed illustration of the IXPE (Imaging X-ray Polarization Explorer) satellite in space. The satellite is shown from a perspective that highlights its large, rectangular solar panel arrays, which are composed of numerous smaller cells. The main body of the satellite is dark and cylindrical, with a large circular aperture visible at the front. A long, thin boom extends from the main body, supporting a complex instrument package at the end, which includes two large, orange-colored cylindrical components. The background is a vibrant, colorful nebula with shades of purple, pink, and red, interspersed with numerous small white stars. In the bottom left corner, a portion of the Earth's blue and green surface is visible, suggesting the satellite's orbit. The overall composition is dynamic and emphasizes the advanced technology of the mission.

Rivelatori telescopio IXPE

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INFN-Pisa

Giornate Studio sui Rivelatori

- ▷ Introduction
- ▷ Measuring X-ray polarization
- ▷ X-ray polarimetry techniques
 - ▷ Bragg diffraction
 - ▷ Photoelectric Effect
- ▷ The Gas Pixel Detector
 - ▷ Description of the components
 - ▷ Assembly process
 - ▷ Event readout and reconstruction
- ▷ The Detector Unit
- ▷ The IXPE satellite

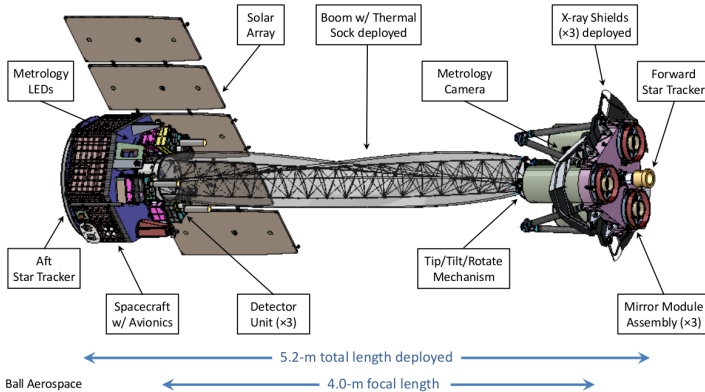
INTRODUCTION

- ▷ **The Imaging X-ray Polarimetry Explorer (IXPE)**
 - ▷ Imaging and polarimetry in the 2–8 keV band
 - ▷ Detectors developed and built at INFN-Pisa
- ▷ Next NASA SMAll EXplorer (SMEX) mission
 - ▷ Launch in early 2021
 - ▷ 2-year mission (baseline), +1 year extension
 - ▷ Equatorial circular orbit at ≥ 540 km altitude
- ▷ International partnership:

 Marshall Space Flight Center PI team, project management, SE and S&MA oversight, mirror module fabrication, X-ray calibration, science operations, and data analysis and archiving  Detector system funding, ground station  Spacecraft, payload structure, payload, observatory I&T	 IAPS  INAF  INFN <small>ISTITUTO NAZIONALE DI ASTROFISICA NATIONAL INSTITUTE FOR ASTROPHYSICS</small> Polarization-sensitive imaging detector systems
	 LASP Mission operations
	 ROMA TRE  Stanford University Scientific theory
	 McGill Science Working Group Co-Chair  MIT Massachusetts Institute of Technology Co-Investigator <small>A12567_151</small>
	

- ▷ X-ray Mirror by NASA/MSFC
- ▷ X-ray Instruments by INFN, IAPS/INAF and ASI
- ▷ Spacecraft, payload structure and integration by Ball Aerospace

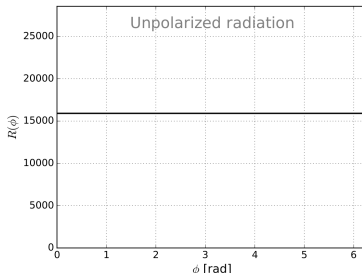
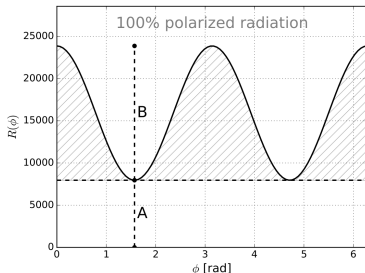
Overview of the observatory



- ▷ Three identical telescopes (redundancy, mitigation of systematic effects, larger acceptance)
- ▷ Conventional Wolter Type I grazing-incidence optics
- ▷ New imaging and polarization-sensitive detector at the focus
- ▷ Extensible boom to save space during launch

- ▷ Polarimetry is already common at many wavelengths
 - ▷ Not really exploited in X-ray
- ▷ Two additional parameters to the phase space:
 - ▷ (linear) polarization degree
 - ▷ polarization angle (phase)
- ▷ Linear polarization whenever there is some “preferred direction” in the system
- ▷ Which means information on many aspects of X-ray sources:
 - ▷ Non-thermal emission processes
 - ▷ Synchrotron radiation
 - ▷ Acceleration phenomena (supernova remnants, pulsar wind nebulae, jets)
 - ▷ Geometry
 - ▷ Photon scattering in aspherical geometries (accretion disks, X-ray reflection nebulae)
 - ▷ Photon propagation in magnetized plasmas (accreting pulsars, magnetars)
 - ▷ Fundamental physics
 - ▷ Quantum electrodynamics (photon propagation in strong magnetic fields)
 - ▷ General relativity (photon propagation in strong gravitational fields)

MEASUREMENT PROCESS



- ▷ Azimuthal modulation around the polarization angle ϕ_0 :

$$R(\phi) = A + B \cos^2(\phi - \phi_0)$$

- ▷ **Modulation factor**: response to 100% polarized radiation:

$$\mu = \frac{R_{\max} - R_{\min}}{R_{\max} + R_{\min}} = \frac{B}{B + 2A}$$

- ▷ Equivalent representation: $R(\phi) = N(1 + m \cos(2(\phi - \phi_0)))$

- ▷ Exercise: calculate μ in this case

- ▷ A real measurement process is thus
 1. Collect a sample of photons
 2. Evaluate ϕ distribution (not as trivial as it may look like)
 3. Extract modulation parameters: amplitude m and ϕ_0

- ▷ Polarization angle is obviously ϕ_0
- ▷ Polarization degree is $p = m/\mu$

- ▷ Notice that p is always positive!
 - ▷ You will get a number even in case of non-polarized beam
 - ▷ Need to understand if it is compatible with zero
 - ▷ Need to understand your sensitivity

- ▷ **Minimum Detectable Polarization (MDP):** (at 99% CL) is the degree of polarization corresponding to the amplitude of modulation that has a 1% probability of being detected by chance

▷ See e.g. M. Weisskopf 2010, <https://arxiv.org/pdf/1006.3711.pdf>

$$MDP = \frac{4.29}{\mu S} \sqrt{\frac{B + S}{T}}$$

S: source rate B: background rate T: observation time

- ▷ In case of negligible background:

$$MDP = \frac{4.29}{\mu \sqrt{ST}} = \frac{4.29}{\mu \sqrt{N}}$$

with N: total number of collected events

- ▷ Commonly used to estimate statistical sensitivity of detectors (and compare them)

- ▷ Let's think positive and assume no bkg and $\mu = 1$
- ▷ How many events to reach a MDP = 1%?
 - ▷ Using $MDP = \frac{4.29}{\mu \sqrt{N}} \Rightarrow N \approx 184000$
- ▷ If we want to be a bit more realistic and assume $\mu = 0.5$
 - ▷ $N \approx 7.36 \times 10^5$
- ▷ Think at these number in comparison with the statistics needed for other measurements
 - ▷ Source detection: ~ 10 events
 - ▷ Spectrum measurement: ~ 1000 events
 - ▷ Polarimetry: $> 10^5$ events
- ▷ What we need is:
 - ▷ a lots of counts – large collecting area, long observation time
 - ▷ a good polarimetric response – maximize μ

- ▷ For each event, define a set of Stokes Parameters

$$i_k = 1 \quad q_k = \cos 2\phi_k \quad u_k = \sin 2\phi_k$$

- ▷ Sum up the parameters for the entire data set:

$$I = \sum i_k \quad Q = \sum q_k \quad U = \sum u_k$$

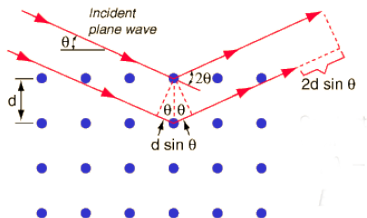
- ▷ Normalize the Stokes parameters:

$$Q = \frac{Q}{I} \quad U = \frac{U}{I}$$

- ▷ Evaluate polarization degree and angle as:

$$p = \frac{2}{\mu} \sqrt{Q^2 + U^2} \quad \phi_0 = \frac{1}{2} \arctan \frac{U}{Q}$$

MEASUREMENT TECHNIQUES



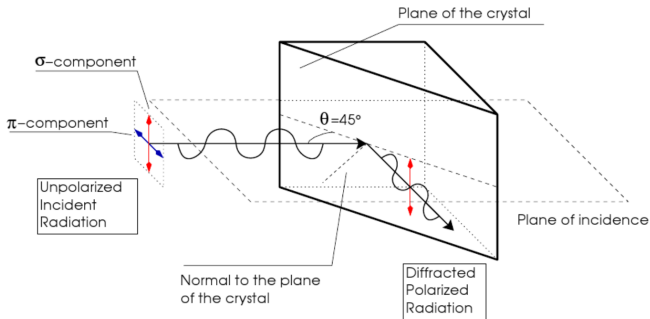
- ▷ X-ray can be diffracted by crystal lattice
- ▷ Constructive interference occurs when

$$n\lambda = 2d \sin \theta$$

or

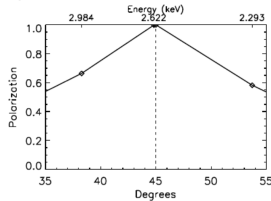
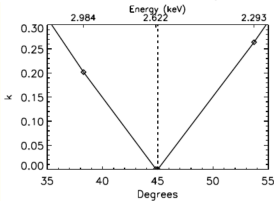
$$E(\theta) = \frac{nhc}{2d \sin \theta}$$

- ▷ Can be used to deflect X-rays, select energy, and select polarization...



- ▷ Diffraction at 45° select one polarization plane
- ▷ Intensity of diffracted beam depends on polarization of incident beam
- ▷ Notice that θ is fixed, so Energy depends only on crystal lattice spacing

Graphite Crystals:

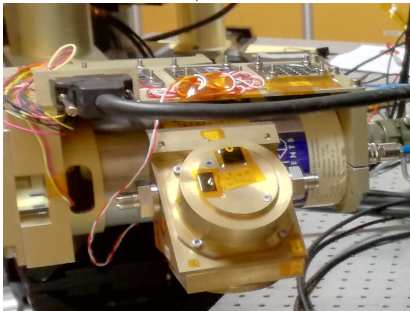


k is the ratio between integrated reflectivity the two polarization components (\parallel and \perp to the diffraction plane)

$$P = \frac{1-k}{1+k}$$

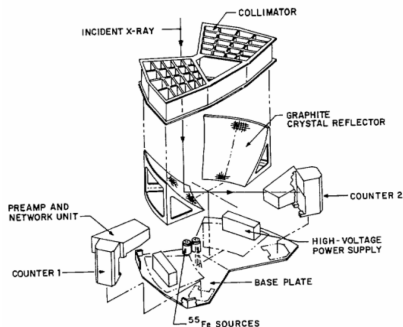
- ▷ Several crystals are suitable for lines in energy range 1–10 keV
- ▷ e.g. graphite works fine at 2.6 keV (and higher order)
- ▷ Mosaic crystals can be used to increase the reflectivity
- ▷ Always keep in mind that real life is complicated:
 - ▷ Try to think at the alignment procedure and the mechanical tolerances that you would need in this kind of setup...

Production of polarized beam:

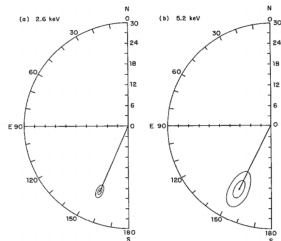
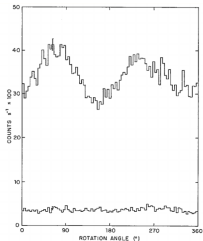
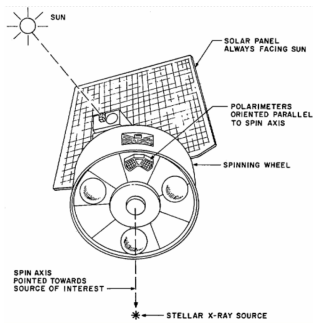


- ▷ For detector calibration
- ▷ Relevant for the IXPE mission

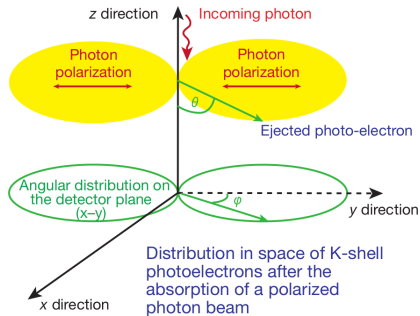
Detection of polarized beam:



- ▷ Used in the first (and only, up to now) polarimeter ever flown
- ▷ One of the instrument on OSO-8 satellite



- ▷ Instrument rotate around pointing axis (spin stabilization of the satellite)
- ▷ Two narrow energy band 2.4–2.8 keV & 4.8–5.6 keV, but $\mu = 0.93$
- ▷ Measurements of the Crab Nebula: $P = 19.22 \pm 0.92\%$
 - ▷ M. C. Weisskopf, ApJL 220 (1978) L117-121



- ▷ Dominant interaction process at low energy (< 10 keV)
- ▷ Distribution of the direction of emission of a K-shell photoelectron 100% modulated for linearly polarized radiation:

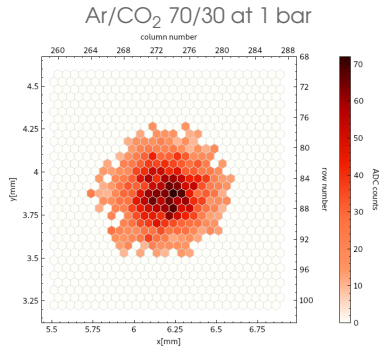
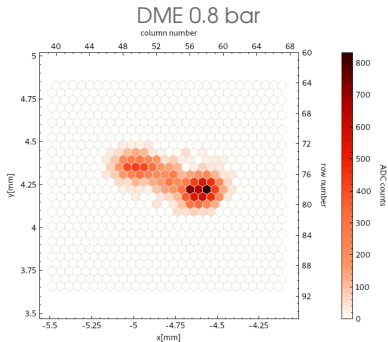
$$\frac{d\sigma_c^k}{d\Omega} \propto Z^5 E^{-\frac{7}{2}} \frac{\sin^2 \theta \cos^2 \phi}{(1 + \beta \cos \theta)^4}$$

- ▷ Need to reconstruct the direction of emission of the photoelectron
- ▷ This is the technique used in IXPE detector

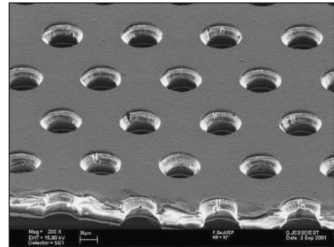
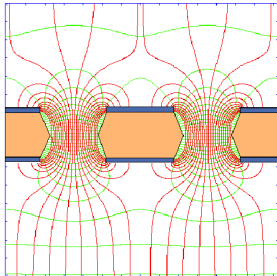
- ▷ The challenge is to be able to measure the initial part of the photoelectron track
 - ▷ Reconstruct the interaction point and the emission direction
- ▷ A good sampling of the photoelectron track is needed
- ▷ Detector granularity should be much smaller than the typical range
 - ▷ Energy dependent effect
 - ▷ In the 1–10 keV energy band typical photoelectron range is a few μm in a solid and a few hundreds μm in a gas
 - ▷ A gas is preferred in this application
 - ▷ There are attempts to use Si detectors at higher energies ($\sim 20 - 100$ keV)
- ▷ Electron scattering at large angle is also a problem
 - ▷ It can smear out the emission direction information
- ▷ Don't forget the highest ionization density is at the end of the track
 - ▷ Bragg peak contains most of the energy
 - ▷ A feature used in track reconstruction

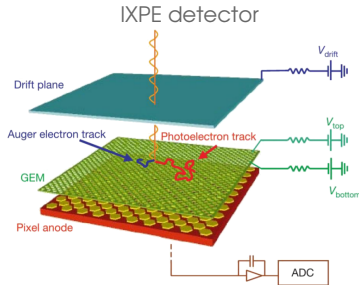
- ▷ Now the tricky part, we have to choose the gas:
 - ▷ Best gas – pure or mixture
 - ▷ Working pressure
 - ▷ Gas gap size
- ▷ Good photon efficiency
 - ▷ Deep absorption gap
 - ▷ High pressure and high Z gas
- ▷ Small diffusion of photoelectron ionization cloud
 - ▷ Diffusion destroys information on ionization point
 - ▷ Gas mixture is fundamental
 - ▷ Cloud drift length is important
 - ▷ Pressure plays a role: diffusion $\sim 1/\sqrt{P}$, but range $\sim P$
- ▷ Interaction must be in k-shell
 - ▷ High-Z element won't work at low energy (e.g. Argon)
- ▷ Auger electron is better than fluorescence
 - ▷ Photoelectron energy is lower than X-ray: $E_{ph} = E_{\gamma} - E_k$
 - ▷ Auger electron is absorbed immediately while fluorescence can travel far or escape

How a 5.9 keV photon looks like in a low (DME) and high (Ar) diffusion gas



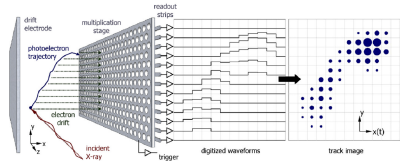
- ▷ Readout pitch as small as possible
 - ▷ Track length a few hundreds μm in the keV energy range
- ▷ 2D readout is necessary: pixel pattern
- ▷ An amplification stage is needed for gas detector and current amplifier
 - ▷ A 1 to 10 keV photon produces between 40 and 400 electron-ion pairs (assuming a $W \sim 25$)
 - ▷ Then divide by the number of pixels involved
- ▷ The amplification stage must preserve position information
 - ▷ e.g. a Gas Electron Multiplier (GEM) with fine pitch
 - ▷ A first track sampling is done by the amplification stage





- ▷ Readout plane orthogonal to photon direction
- ▷ Absorption gap limited by diffusion
- ▷ Imaging and polarimetry at the same time
- ▷ Azimuthally asymmetric

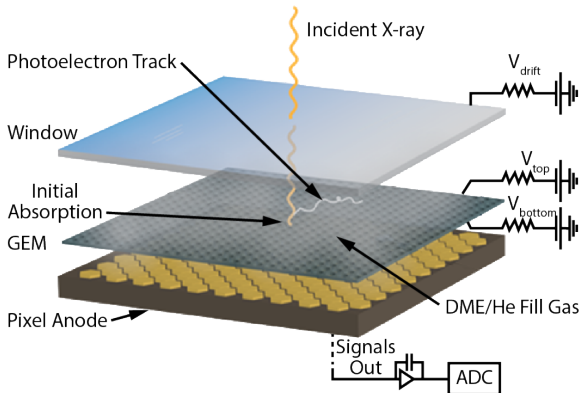
TPC concept



- ▷ Readout plane parallel to photon direction
- ▷ Drift to the side allows for a deep absorption gap (efficiency)
- ▷ Imaging only in one direction (no absorption time available)
- ▷ Highly azimuthally asymmetric

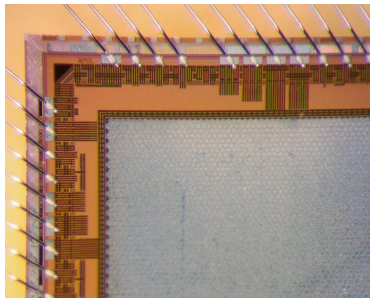
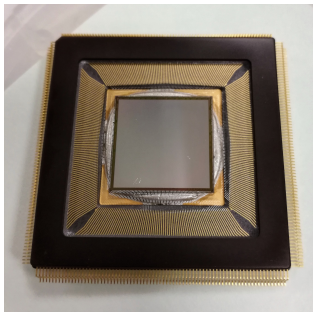
THE GAS PIXEL DETECTOR

The Gas Pixel Detector (GPD)



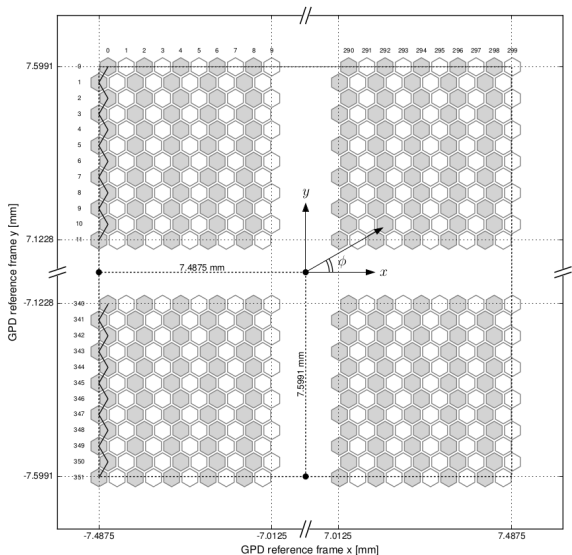
- ▷ Gas gap for X-ray absorption
 - ▷ 1 cm of DME at 800 mbar
- ▷ Signal amplification via a Gas Electron Multiplier (GEM)
- ▷ Finely pixelized ASIC as readout anode
- ▷ Designed for energy range $\sim 2 - 8$ keV

The core of the detector: the ASIC



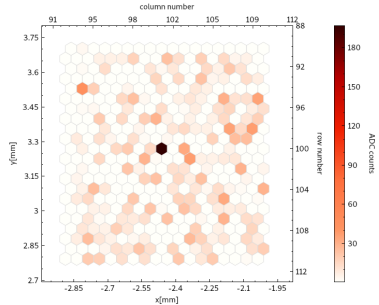
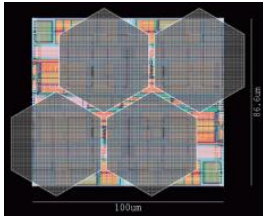
Properties

Pixels organization	300×352 pixels in hexagonal pattern
Pixel pitch	50 μm
Active area	15×15 mm ²
Shaping time	4 μs
Pixel Noise	~ 50 electrons ENC
Trigger	internal, with definition of a region of interest
Output	analog (external ADC required)
Technology	CMOS 0.18 μm

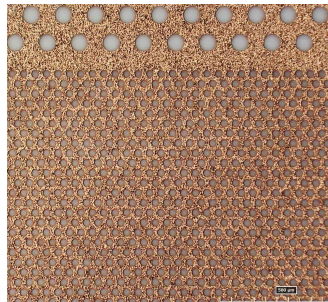
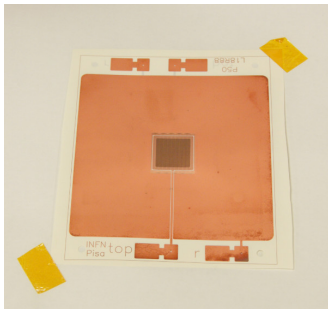


Each pixel has:

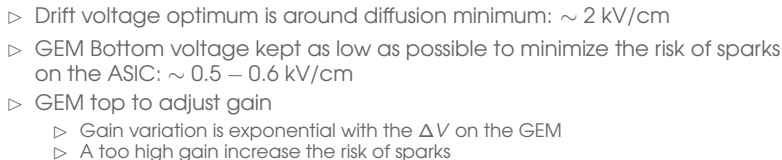
- ▷ Hexagonal electrode, top layer is metal
- ▷ A charge-sensitive amplifier followed by a shaping circuit
- ▷ Address by its column and row, (0,0) on top-left

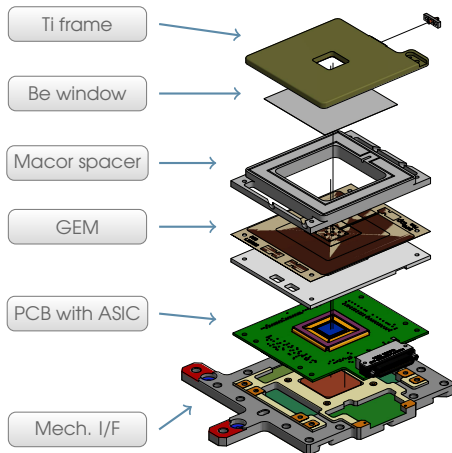


- ▷ Pixels are grouped in 2×2 minicluster to contribute to a single trigger with dedicated shaping amplifier
- ▷ Single trigger threshold for all the ASIC
- ▷ Pixels can be individually masked to the trigger
- ▷ Autonomous definition of a square region-of-interest (ROI) around the triggering miniclusters
 - ▷ With a margin of ~ 10 pixels
- ▷ Serial readout of the pixels inside the ROI
 - ▷ A clock is sent to the ASIC
 - ▷ At each cycle the next pixel is connected to the analog output buffer
 - ▷ An external ADC read the charge of the pixel



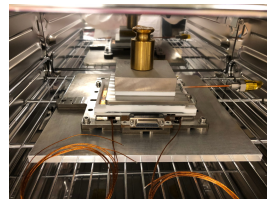
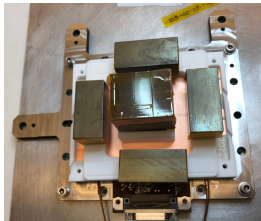
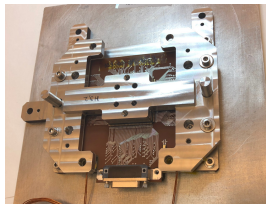
- ▷ Produced by RIKEN and SciEnergy in Japan
- ▷ Hexagonal hole pattern, with 50 μm pitch, 50 μm thick
- ▷ Active size matching ASIC + large guard ring for uniform drift field
- ▷ Liquid crystal polymer (LCP) insulator (laser etching technique)
- ▷ Mask alignment at a few μm level



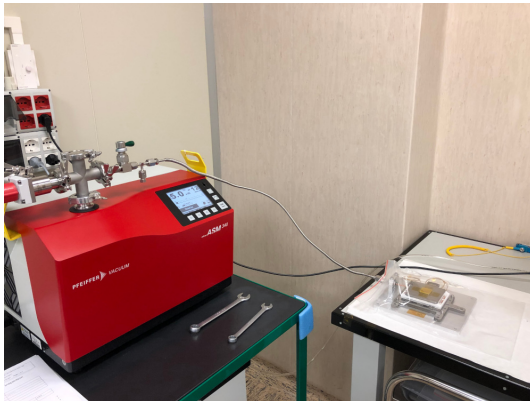


- ▷ Sealed detector
 - ▷ No gas system needed
- ▷ Ceramic parts for gas cell and GEM support
 - ▷ Low outgassing, for space application and gas purity
- ▷ A Ti frame acts as “drift” electrode
- ▷ X-ray window in Be, 50 μm thick
- ▷ ASIC in a standard package mounted on a custom PCB
 - ▷ Commercial ceramic package
 - ▷ Space compatible PCB
- ▷ A Ti frame for mechanical and thermal interface

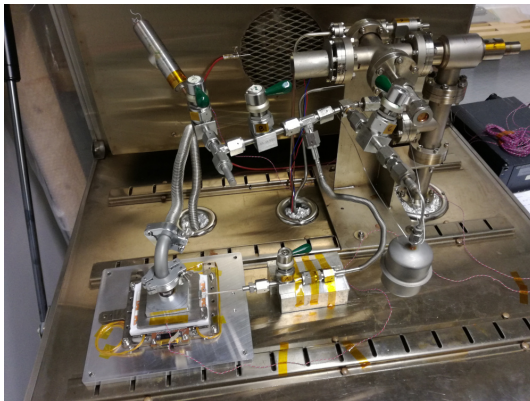
Gluing everything together



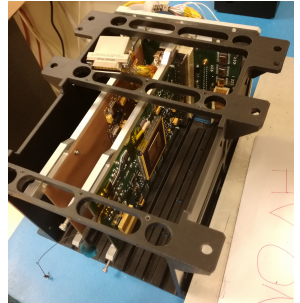
- ▷ Use space compatible adhesive
 - ▷ Service temperature $> 100\text{ }^{\circ}\text{C}$ (detector bake-out)
- ▷ Find the right glue pattern (for a good sealing)
- ▷ Find the right gluing sequence (avoid thermal stress)
- ▷ Keep items in place during glue curing



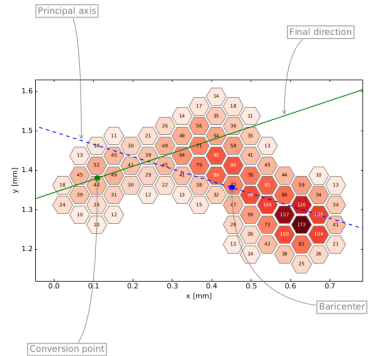
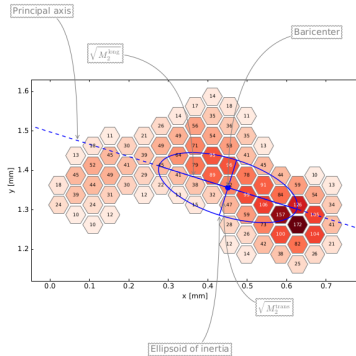
- ▷ Detector is sealed and has to last the entire mission without refilling
- ▷ Severe requirement on leak rate: $< 1 \cdot 10^{-9}$ mbar l/s
- ▷ All GPD are tested with He leak detector after assembly



- ▷ Detector bake-out and filling is done in a dedicated facility at OIT in Finland
- ▷ A 2 weeks bake-out at 100 °C to clean the gas chamber
- ▷ Filling with DME at 0.8 bar is done in the same facility
- ▷ Finally GPD is sealed by crimping the filling tube

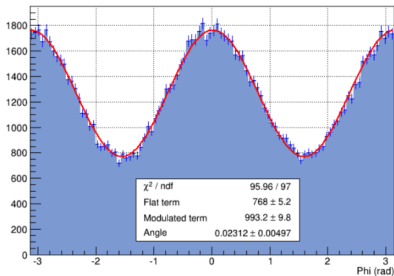


- ▷ Four PCBs in a dedicated housing:
 - ▷ Data Acquisition board (DAQ)
 - ▷ Low Voltage Power Supply, Board (LVPS)
 - ▷ High Voltage Supply Board (HVPS)
 - ▷ Back Plane (BP)
- ▷ FPGA based DAQ, with a 14-bit ADC for GPD data
- ▷ Two custom digital interfaces for communication:
 - ▷ Command and Control Interface (CCI)
 - ▷ Science Data Interface (SDI)
- ▷ Event timing via 1-PPS (from spacecraft GPS) and a 1 MHz clock
- ▷ Dedicated mechanical frames provide stiffness and thermal control

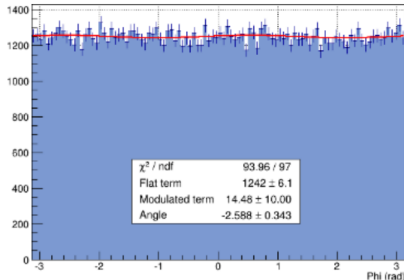


- ▷ Event by event reconstruction
- ▷ Iterative moment analysis to reconstruct relevant information
 - ▷ Interaction point: **imaging**
 - ▷ Photoelectron direction: **polarimetry**
 - ▷ Trigger output: **timing**
 - ▷ Pixel charge content: **spectroscopy**

polarized source

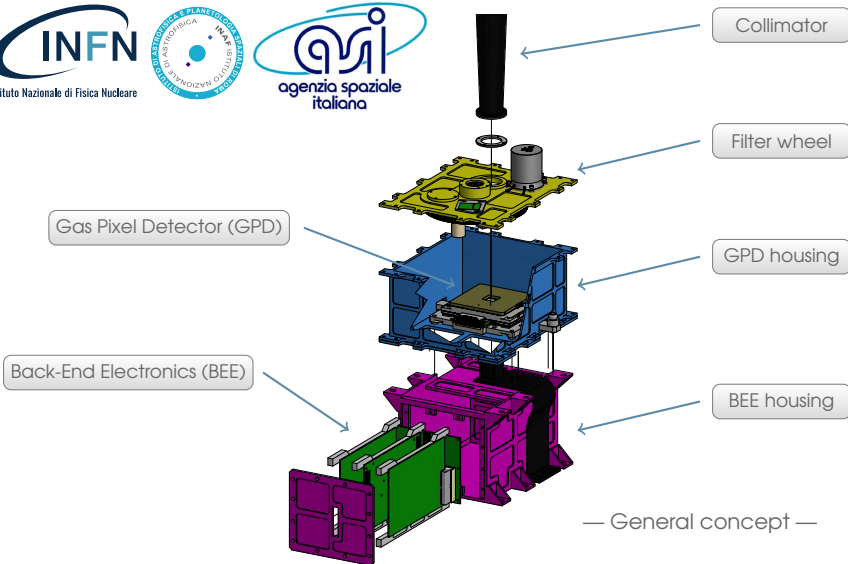


unpolarized source



- ▷ **Modulation factor:** 0.2 (0.7) at 2 (8) keV
 - ▷ Stability over ~ 3 years demonstrated with a sealed detector
- ▷ $\sim 90 \mu\text{m}$ **spatial resolution** at 5.9 keV, measured (\ll track length)
 - ▷ Good match for a ~ 25 arcsec-type X-ray optics with ~ 4 m focal length
- ▷ $< 20\%$ **energy resolution** (FWHM) at 5.9 keV
 - ▷ Enough for spectrally-resolved polarimetry (in a few energy bins) when statistics allow it
- ▷ μs -type **time resolution**
 - ▷ More than adequate for the shortest time scales of interest

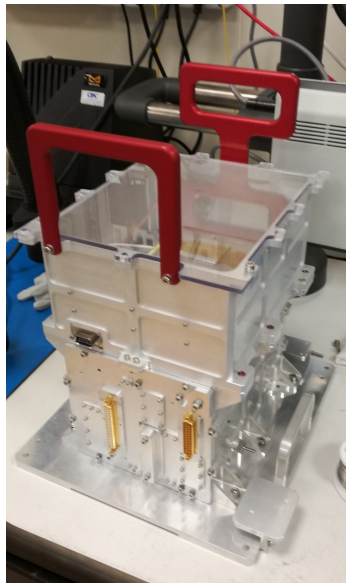
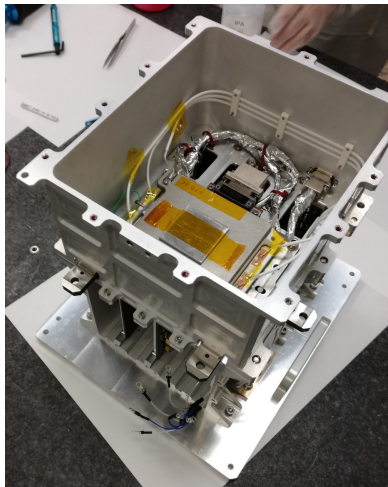
THE DETECTOR UNIT



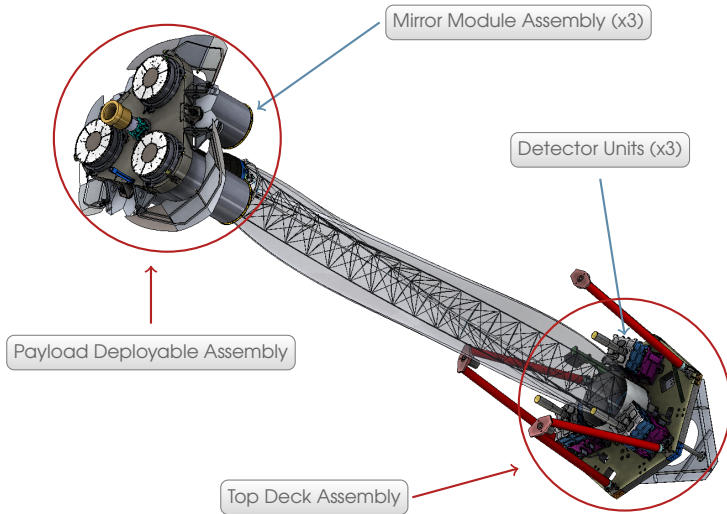
— General concept —

The Detector Unit (DU)

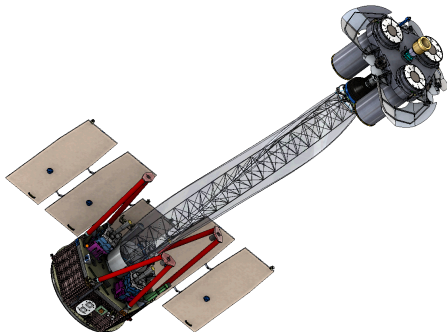
First assembly of the Engineering Model



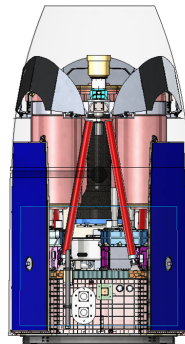
THE IXPE SATELLITE



Deployed

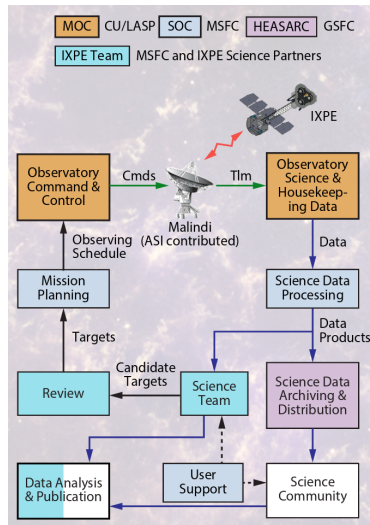


Stowed



- ▷ Optical boom to be deployed after launch to extend the optics at the right position
- ▷ Satellite 3-axis stabilized, GPS positioning and star-tracker for pointing
- ▷ S-band communication
- ▷ Launch in stowed configuration, compatible with Pegasus XL fairing

- ▷ Point-and-stare observations of known target
- ▷ S-band downlink via ground station (Malindi)
- ▷ Observation plan for the first year almost ready
- ▷ Open to community requests in the second year
- ▷ Data are immediately public (after downlink and validation)



THE END